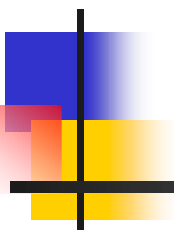


Fault-tolerant Control System Design and Analysis



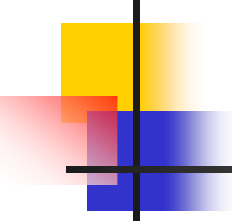
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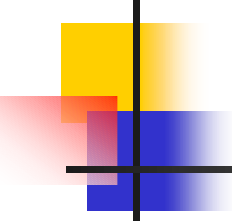
London, Ontario N6A 5B9

Canada



Outline of the presentation

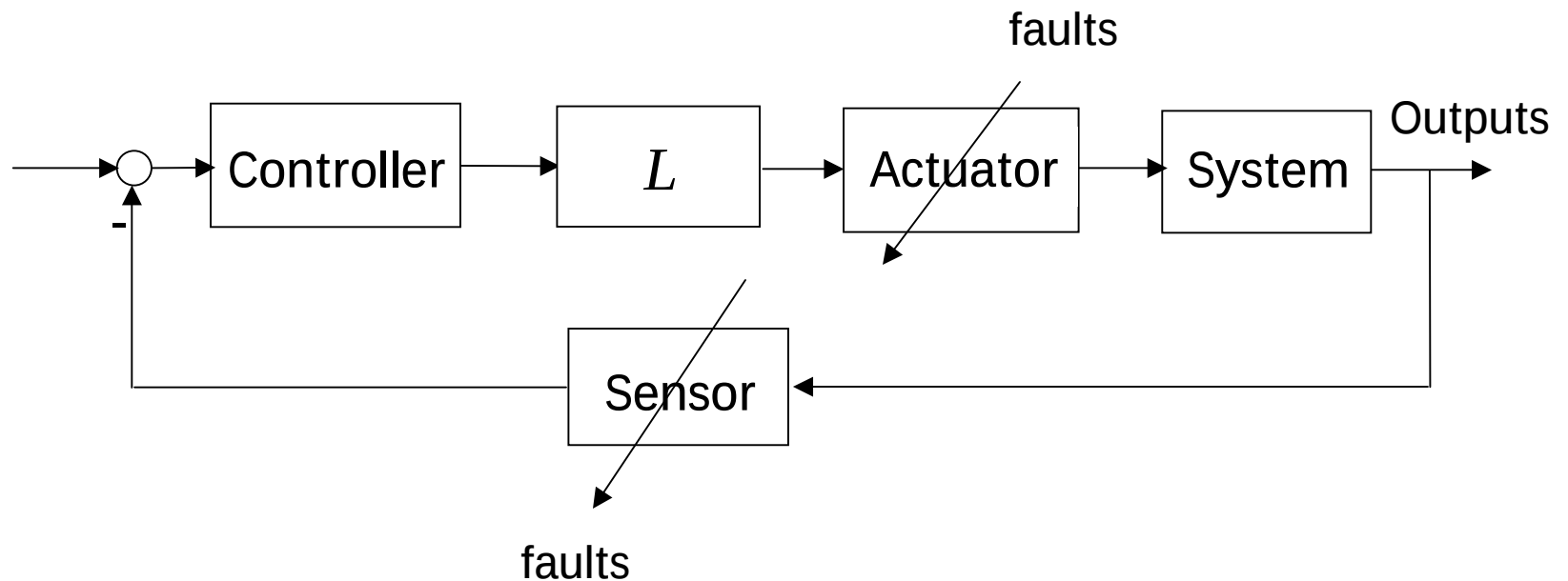
- Overview of two approaches to fault-tolerant control system design and analysis
- Redundancy in fault-tolerant control systems
- Trade-offs among redundancy, performance and integrity
- An example of passive fault-tolerant control design
- An example of active fault-tolerant control design
- Some open problems



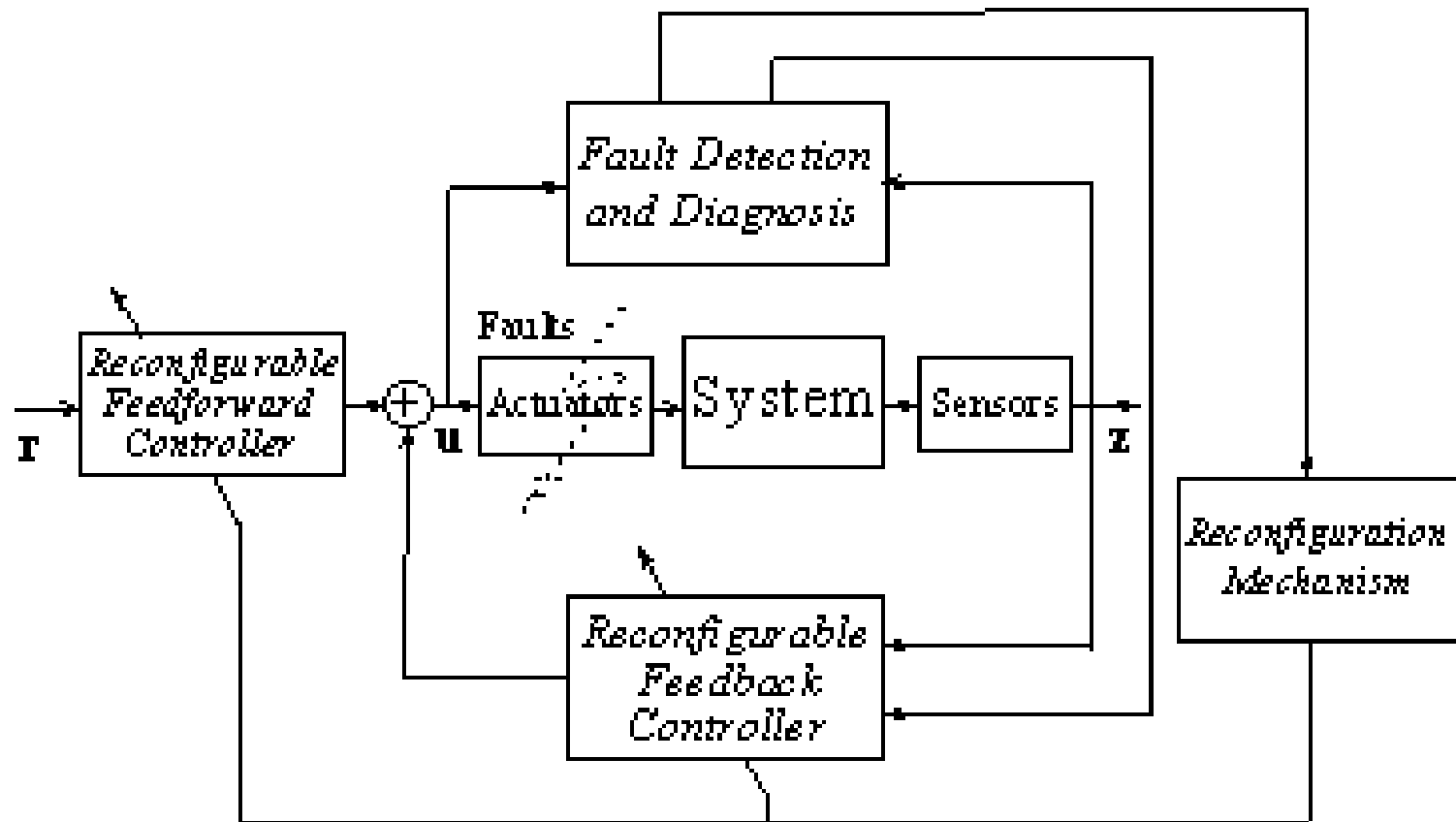
Fault-tolerant control: An overview

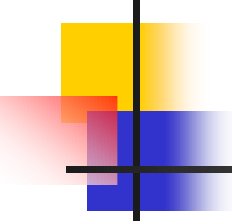
- Passive fault-tolerant control systems
 - Robust fixed structure controller
 - Faults have been considered at the controller design stage
- Active fault-tolerant control systems
 - Explicit fault detection/diagnosis schemes
 - Real-time decision-making and controller reconfiguration
- The key to any fault-tolerant control system
 - Redundancy

Passive fault-tolerant control systems



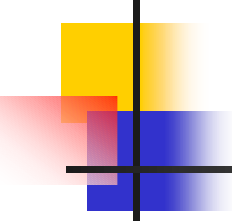
Active fault-tolerant control systems





Features and limitations

- **Passive fault-tolerant control systems**
 - Simple to implement
 - Difficult to account for large number of fault scenarios
 - Unable to deal with unforeseen faults
- **Active fault-tolerant control systems**
 - Potentially be able to deal with a large number of fault scenarios
 - Can deal with certain number of unforeseen faults
 - More complex to implement
 - Real challenge is real-time decision-making



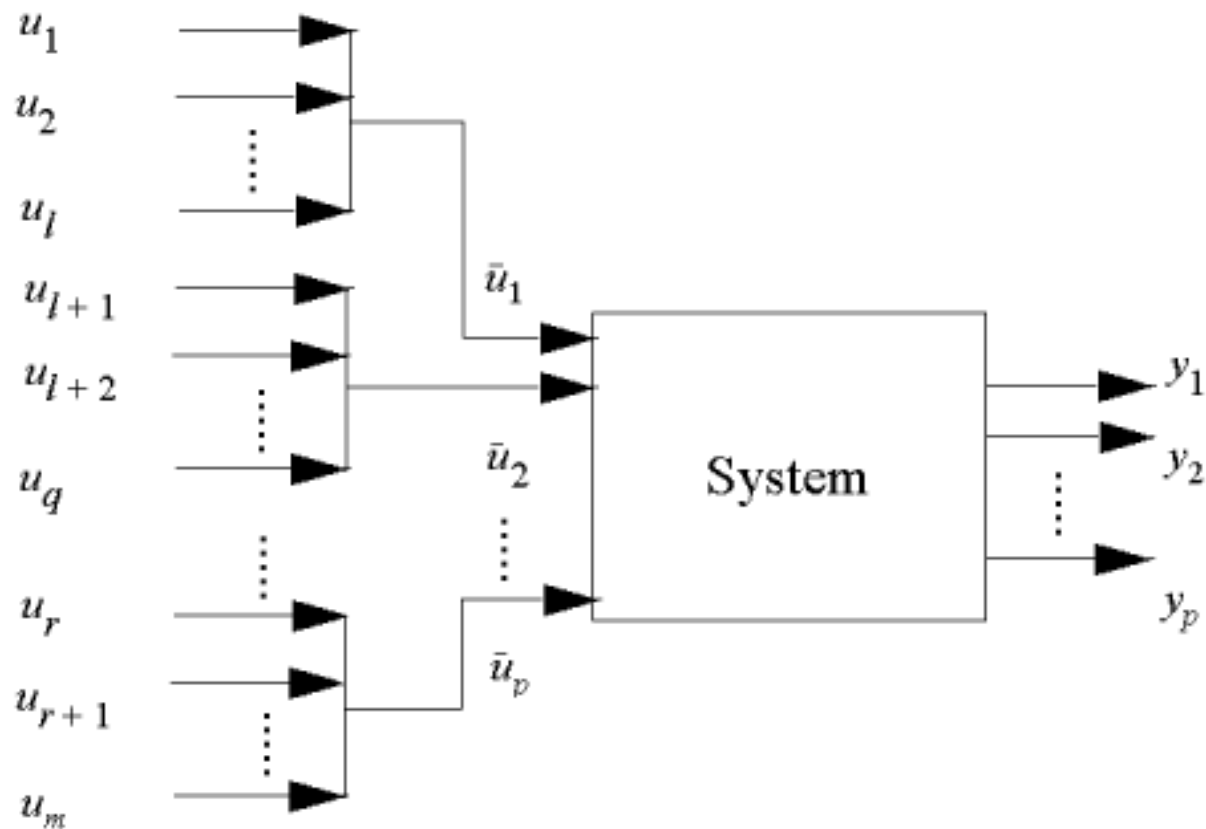
Redundancies

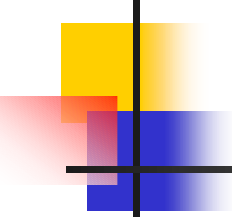
- Actuator redundancies
 - Multiple physical actuators
 - They usually act on the system at different locations

- Sensor redundancies
 - Multiple physical sensors
 - They usually measure the same physical quality

- Analytical redundancies
 - Rely on mathematical models (FDI)

Actuator redundancies





Actuator Redundancies

For a multi-input linear system with the following state space representation:

$$\begin{aligned}\dot{x} &= Ax + Bu \\ y &= Cx\end{aligned}$$

where $x \in \mathbb{R}^{n \times 1}$, $y \in \mathbb{R}^1$ are the system state and the output, respectively. The system and the output matrices, A and C , are assumed to have appropriate dimensions. The input matrix

$B \in \mathbb{R}^{n \times p}$ can be represented by $B = \begin{bmatrix} b_1 & b_2 & \dots & b_p \end{bmatrix}$ with each column being

$b_i \in \mathbb{R}^{n \times 1}$ $1 \leq i \leq p$. The system input vector associated with the multiple actuators is

given by $u = \begin{bmatrix} u_1 & u_2 & \dots & u_p \end{bmatrix}^T$. Three types of redundancy can be defined.



Definitions of actuator redundancies

Definition 2.1:

*The system of (EQ 1) is said to have **(p-1)** degree of actuator redundancy, if the pair (A, b_i) is completely controllable $\forall i \quad (1 \leq i \leq p)$.*

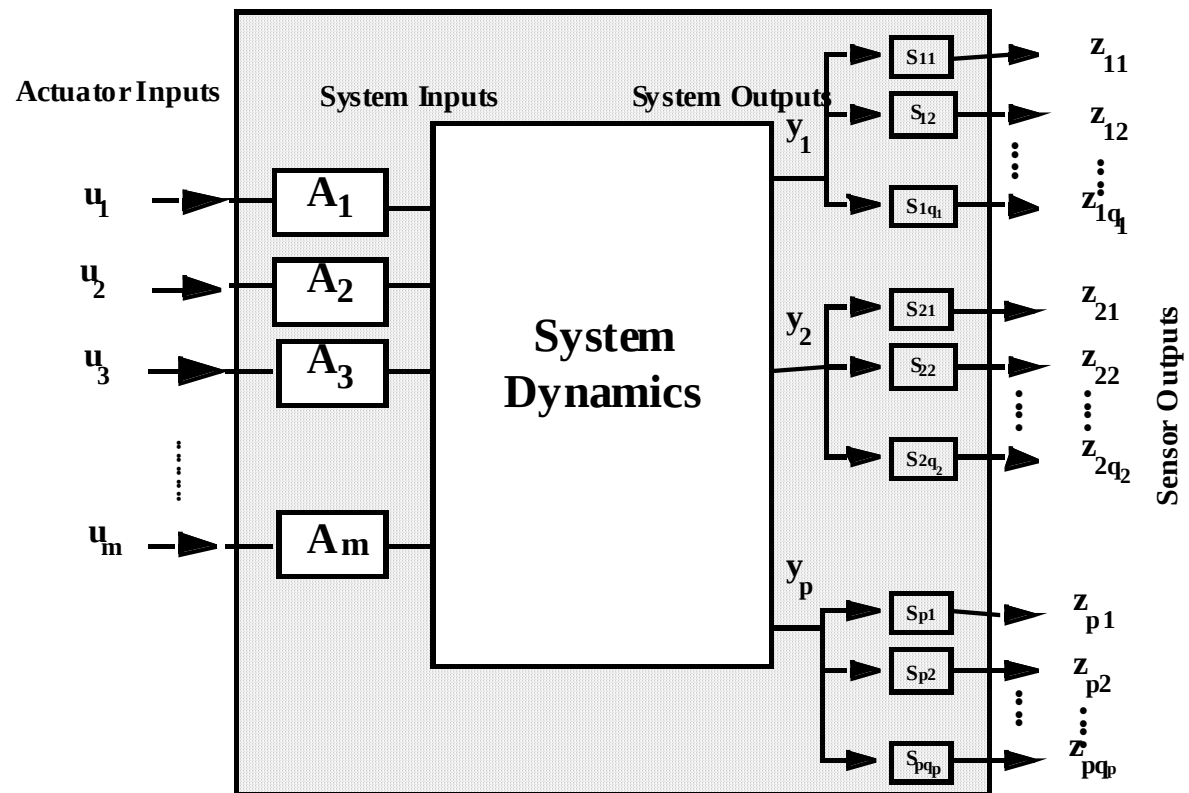
Definition 2.2:

*The system of (EQ 1) is said to have **(p-1)** degree of **non-uniform** actuator redundancy, if (A, b_i) is completely controllable $\forall i \quad 1 \leq i \leq p$ and the $\text{Rank}[B] = p$.*

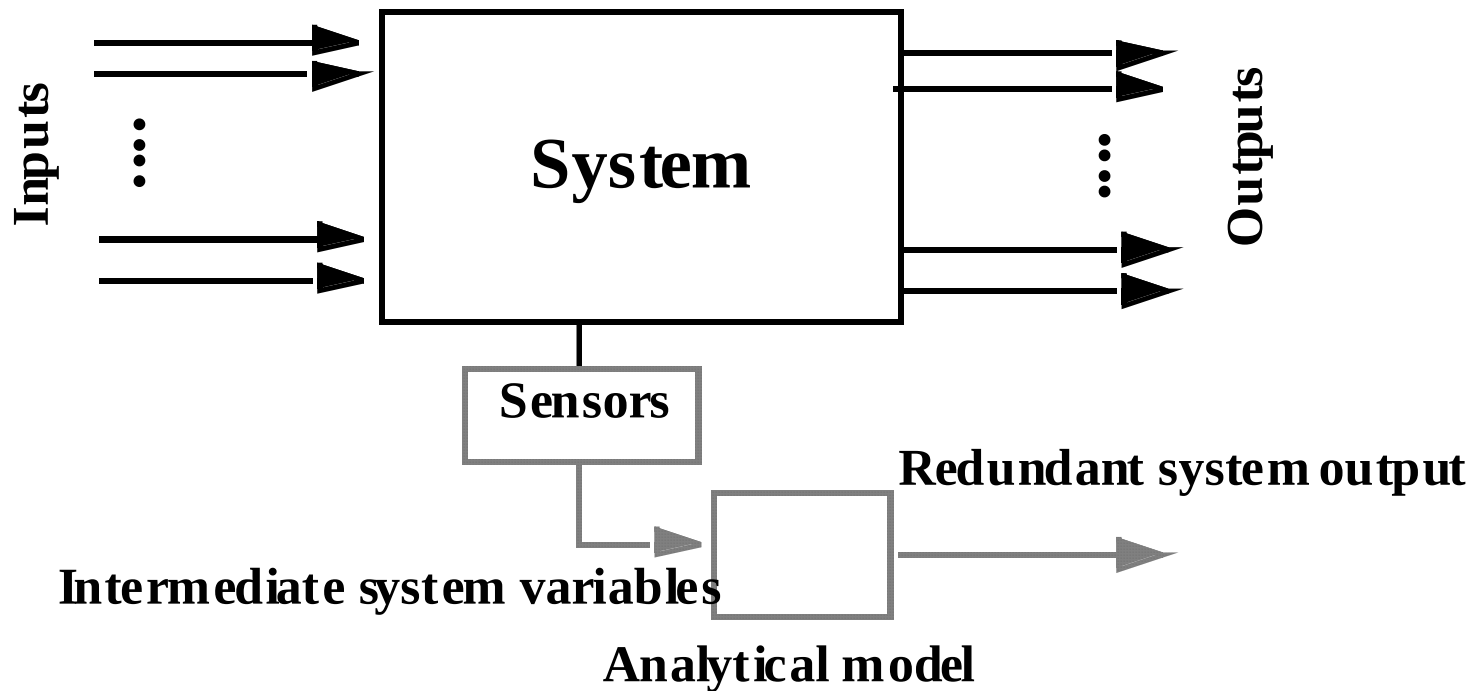
Definition 2.3:

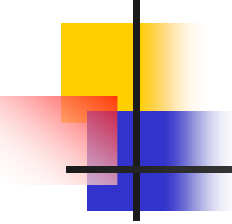
*The system of (EQ 1) is said to have **(p-1)** degree of **uniform** actuator redundancy if (A, b_i) is completely controllable $\forall i \quad 1 \leq i \leq p$ and the $\text{Rank}[B] = 1$.*

Sensor redundancies



Analytical Redundancies





Performance trade-offs

Three main factors to consider in any fault-tolerant control system design:

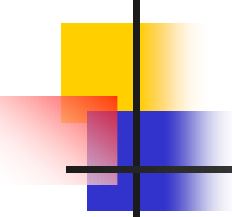
- System Integrity (safety requirements)
- Performance (design specifications)
- Redundancy (physical and financial constraints)

Problem: How to design a control system, under a given degree of redundancy such that the integrity of the system is guaranteed and the performance is satisfactory.

Issue 1: System integrity should always be maintained

Issue 2: Faults should result in reduction of the degree of redundancy first

Issue 3: One should consider performance degradation with available redundancies.



Example of passive fault-tolerant control system

The system used in this example represents a bank-angle control system for a jet transport aircraft flying at the speed of 0.8 Mach, and the attitude of 40,000 ft. There are two manipulated variables: the aileron, and the rudder. The variable being controlled is the bank-angle of the aircraft.

The transfer function matrix for this system is given as follows:

$$G(s) = \begin{bmatrix} \frac{1.1476s^2 - 2.0036s - 13.7264}{s^4 + 0.6358s^3 + 0.9389s^2 + 0.5116s + 0.0037} & \frac{10.7290s^2 + 2.3169s + 10.237}{s^4 + 0.6358s^3 + 0.9389s^2 + 0.5116s + 0.0037} \end{bmatrix}$$

To convert the non-uniform actuator redundancy to a uniform one, the following dynamic pre-compensator is used:

$$D(s) = \text{diag} \left[\frac{(s + 0.108)^2 + 0.9708^2}{s^2}, \frac{(s - 4.4399)(s + 2.6940)}{s^2} \right]$$



Description of system

With such a pre-compensator, the augmented system can be represented in the following state-space form:

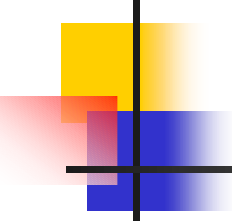
$$\dot{x}_a(t) = \begin{bmatrix} 0.0 & 1.0 & 0.0 & 0.0 & 0.0 & 0.0 \\ 0.0 & 0.0 & 1.0 & 0.0 & 0.0 & 0.0 \\ 0.0 & 0.0 & 0.0 & 1.0 & 0.0 & 0.0 \\ 0.0 & 0.0 & 0.0 & 0.0 & 1.0 & 0.0 \\ 0.0 & 0.0 & 0.0 & 0.0 & 0.0 & 1.0 \\ 0.0 & 0.0 & -0.0037 & -0.5116 & -0.9389 & -0.6358 \end{bmatrix} x_a(t) + \begin{bmatrix} 0.0 & 0.0 \\ 0.0 & 0.0 \\ 0.0 & 0.0 \\ 0.0 & 0.0 \\ 0.0 & 0.0 \\ 1.1476 & 10.7290 \end{bmatrix} \begin{bmatrix} l_1 & 0 \\ 0 & l_2 \end{bmatrix} u_a(t)$$

and the output equation becomes:

$$y(t) = [-11.4125 \quad -4.2488 \quad -11.3838 \quad -1.53 \quad 1 \quad 0] x_a(t)$$

The following state feedback gain matrix is obtained:

$$K = \begin{bmatrix} 1.525 \times 10^{-2} & 0.14012 & 0.5257 & 1.05 & 1.1154 & 0.9106 \\ 2.256 \times 10^{-3} & 2.051 \times 10^{-2} & 7.736 \times 10^{-2} & 0.1211 & 0.1585 & 0.1241 \end{bmatrix}$$



Control system performance

TABLE 1. The eigenvalues of the closed-loop system under three modes of operation

Normal Operation	Aileron Failure	Rudder Failure
$-0.2937+j0.6001$	$-0.2606+j1.0475$	$-0.5052+j0.9191$
$-0.2937-j0.6001$	$-0.2606-j1.0475$	$-0.5052-j0.9191$
-0.3468	$-0.1068+j0.3094$	$-0.2040+j0.4738$
-1.0782	$-0.1068-j0.3094$	$-0.2040-j0.4738$
-0.5	-0.7621	$-0.2745+j0.0858$
-0.5	-0.1840	$-0.2745-j0.0858$

Control system performance

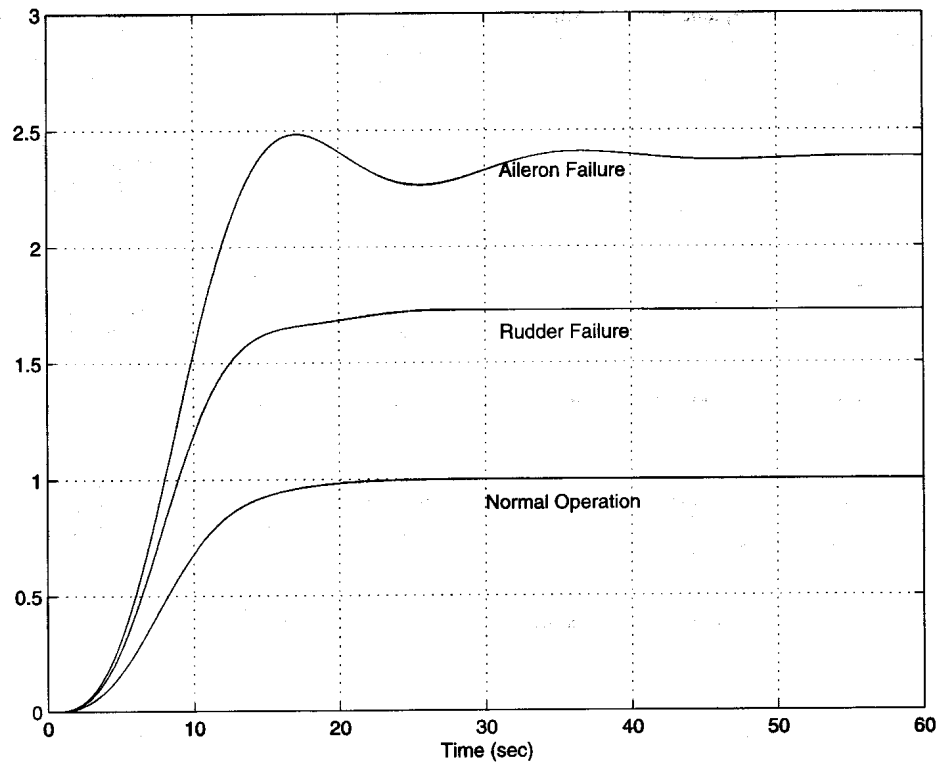
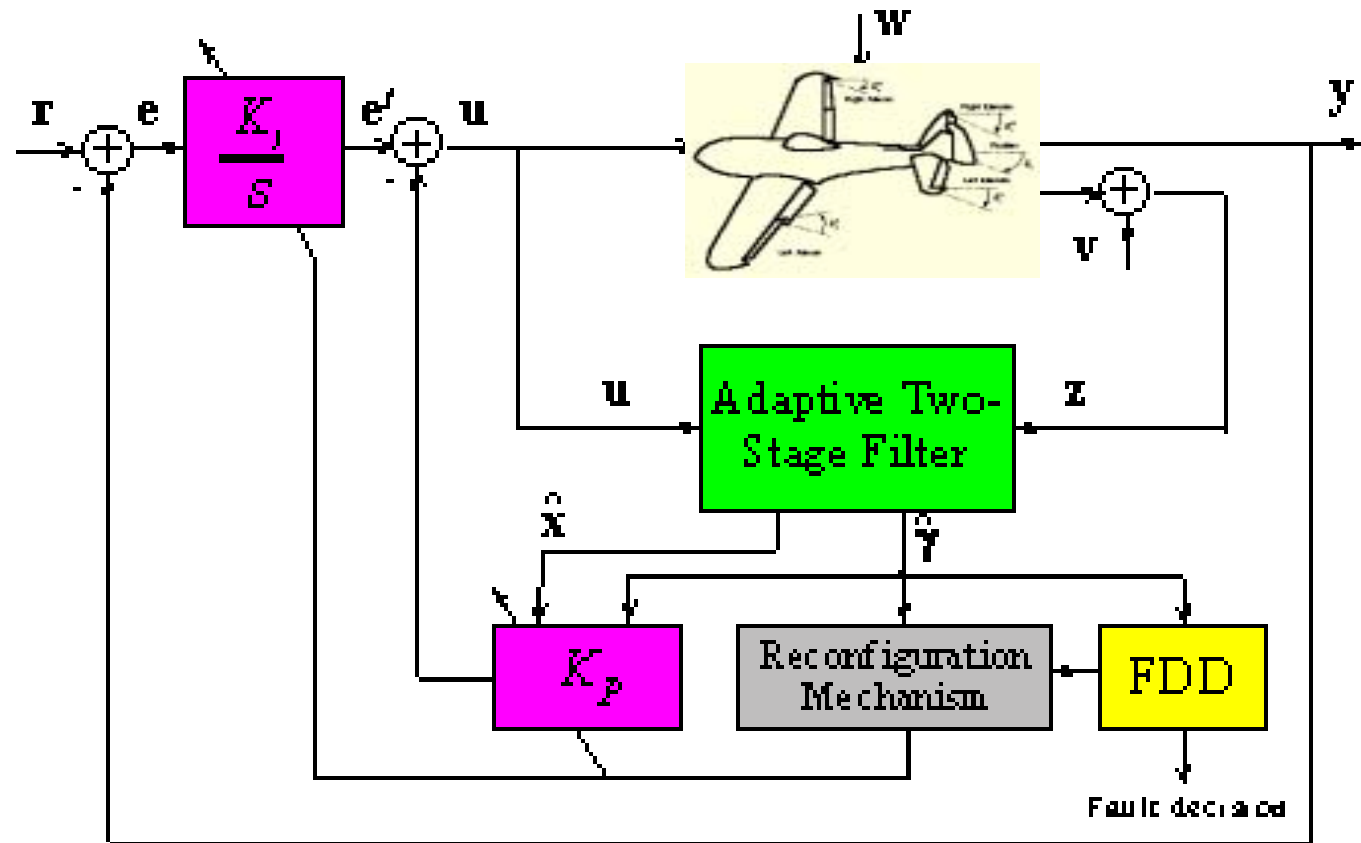
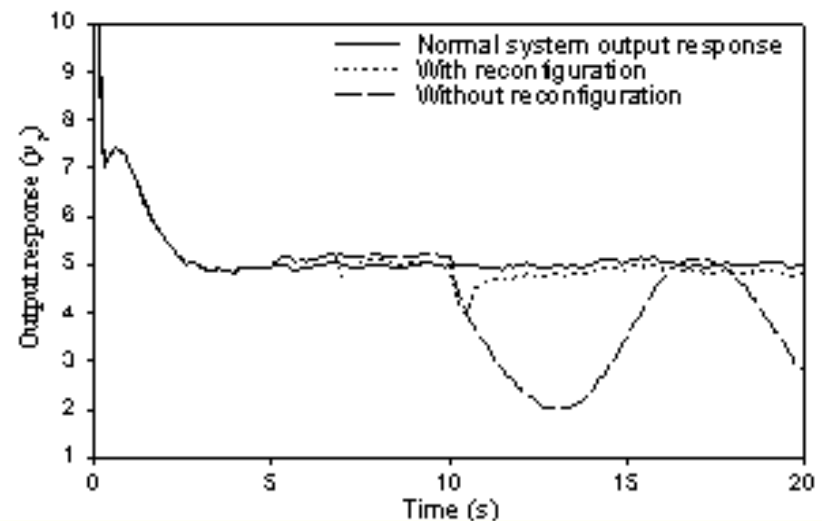
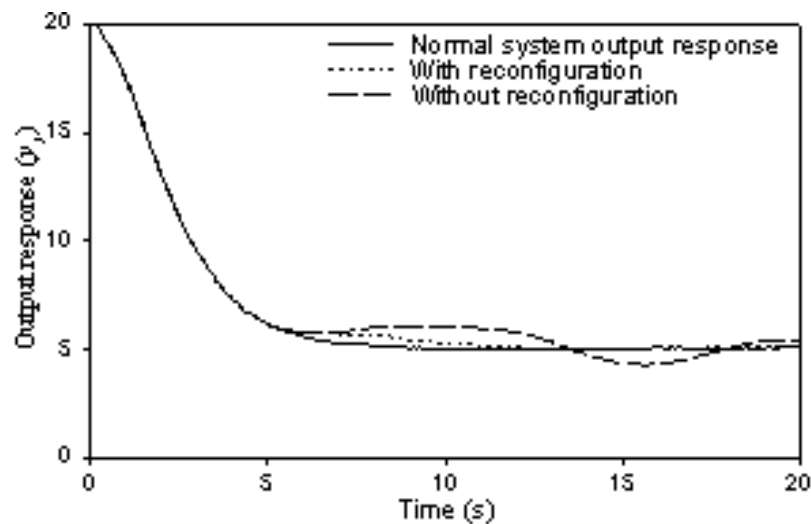


Fig. 3. Step responses of the system for different actuator operating modes.

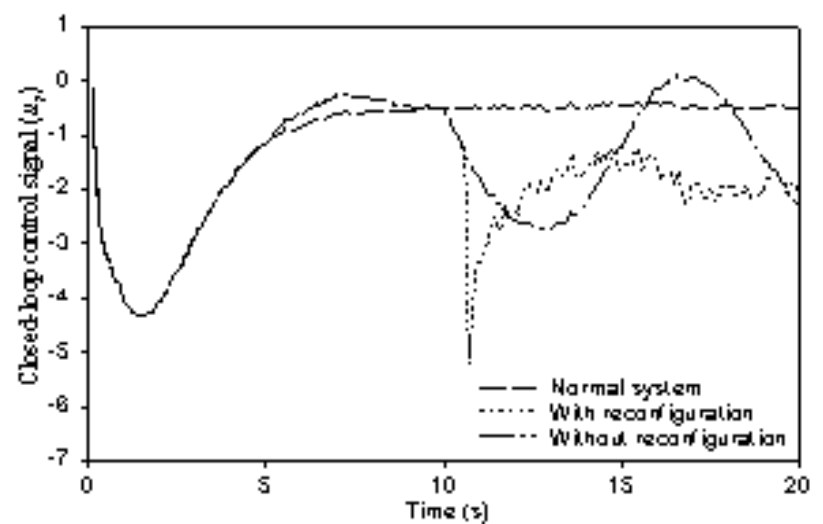
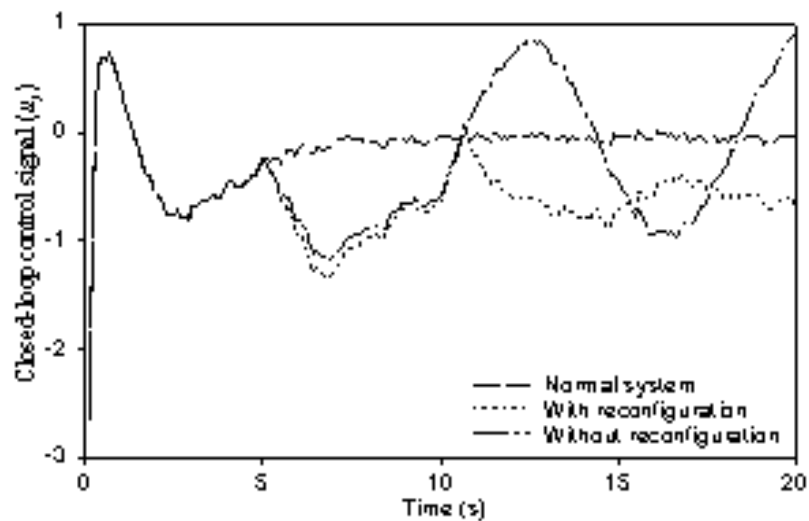
Example of active fault-tolerant control system



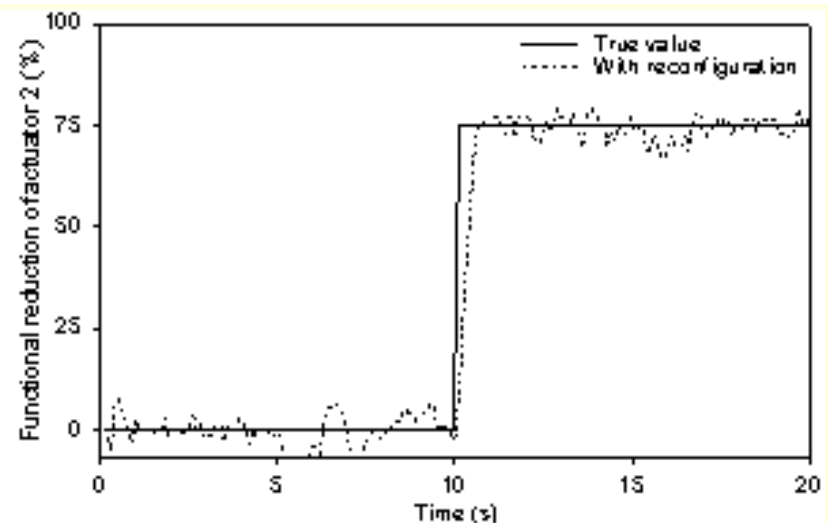
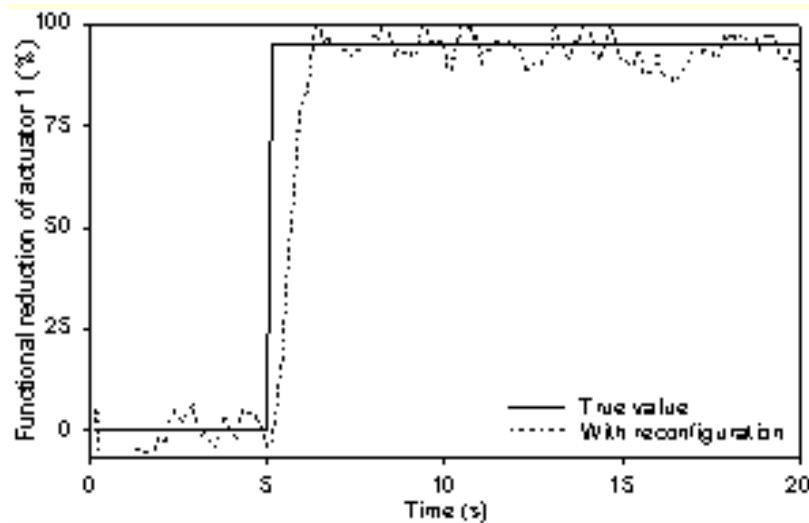
Simulation results



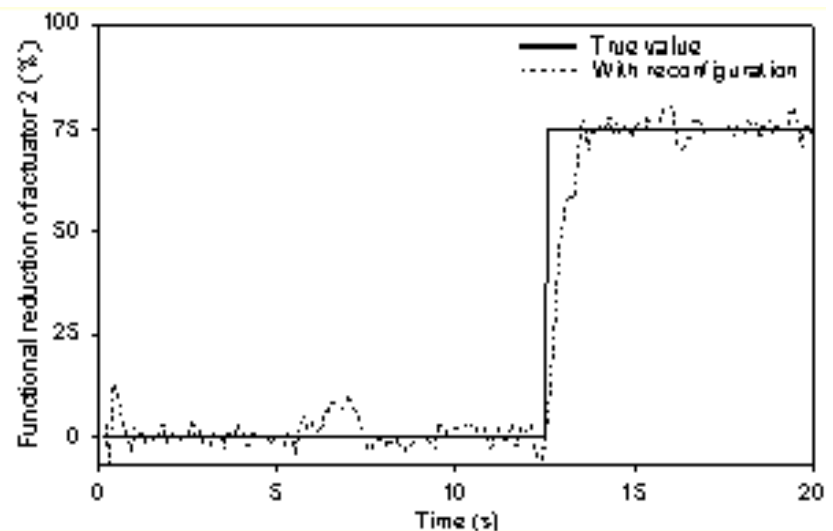
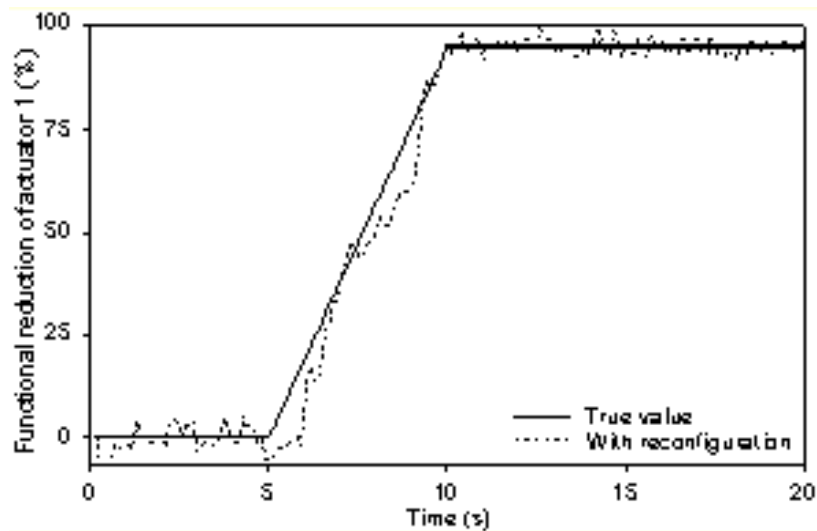
Simulation results

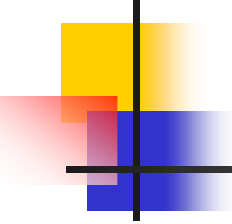


Simulation results



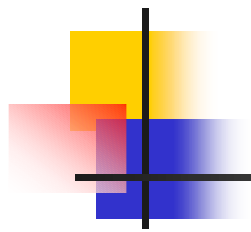
Simulation results





Some open problems

- Reliability analysis of fault-tolerant control systems
- Stability analysis of fault-tolerant control systems
- Graceful performance degradation
- Integration of passive and active approaches
- Industrial applications of fault-tolerant control system technologies



Thank You !